

a) In level flight, $V = \left(\frac{2W/S}{\rho C_L} \right)^{1/2}$

So V_{min} occurs at $C_{Lmax} = 0.8 \rightarrow V_{min} = \left(\frac{2W/S}{\rho C_{Lmax}} \right)^{1/2} = 4.56 \text{ m/s}$

b) $D_0 = \frac{1}{2} \rho V^2 C D A_0$

$D_p = \frac{1}{2} \rho V^2 c_d S$

$D_i = \frac{(W/b)^2}{\frac{1}{2} \rho V^2 \pi e}$

$D = D_0 + D_p + D_i$ (plot)

c) Maximum speed V_{max} occurs when drag reaches max available thrust,

$D(V_{max}) = T_{max}$

or $\frac{1}{2} \rho [C D A_0 + c_d S] V^2 + \frac{(W/b)^2}{\frac{1}{2} \rho \pi e} \cdot \frac{1}{V^2} - T_{max} = 0$
 $A V^4 - T_{max} V^2 + C = 0$

$A = 0.0066, T_{max} = 1.5, C = 1.551$

$\rightarrow V^2 = \frac{1}{2A} [T_{max} + \sqrt{T_{max}^2 - 4AC}] = 226.2$

$V = 15.04 \text{ m/s}$

Could also obtain graphically

